#### **Analysis of Multistorey Frame for Various Loadings**

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#### **ABSTRACT**

We know multistorey structure portal frames are most common in all countries. All metropolitan cities and upcoming cities and major industries use multistorey frames in their structural systems. A Frame of G+3, one bay with point load and G+5, one bay has been take for study, manually the point loads and uniformly distributed load of horizontal elements have been done. Manually analysis has been done using Stiffness Matrix Method of Deflection, Reaction, Banding Moments and Shear Forces. The Multistorey same frames also create in STAAD Pro, and out-put such as Deflection, Reaction, Banding Moments and Shear Forces have been calculated and a comparison between Manual Calculation and STAAD Pro. Software Calculations has been made.

#### Keywords: Stiffness matrix method, Deflection, Reaction, Banding Moments and Shear Forces I.INTRODUCTION

The Multistorey Frame Analysis as we know it today, evolved over several years in this world. During this time many countries devolved but "India" is still one of devolving countries. The main aim of manually calculation and comparison with STAAD Pro is to provide a sustainable cost as compeer to STAAD Pro .The Excel sheet which is based on manually calculation it will easy to operate for most common portal frames. Also this paper will present of comparison and deference of results.

This comparison will helpful to crest and examples of multistory frames with manually calculation. The studies on stiffness matrix method for portal frames up to G+3, one bay and G+5 one bay for more research works of frames Deflection of elements, Reactions on foundation, banding moment, shear force of elements.

#### II.STIFFNESS MATRIX METHOD

The stiffness matrix is a matrix method and a method of analysis to find the member forces and displacements of joints in structure. Under the influence of loads, and support movement, this method can be used to find the displacements and internal forces.

The analysis of both skeletal structures and non-skeletal structures can be done by this method. Beams, plane and space frames and trusses, and grids are skeletal structure. Plates, shells, and 3-D solids are non-skeletal structure.

The stiffness matrix method has a sequential procedure which contains following six steps:

#### **III.MODELING**

- 1.1) First of all model the structure with members and joints
- 1.2) Specify the members and joints of the structure
- 1.3) Recognize all the components of join displacement which are unknown

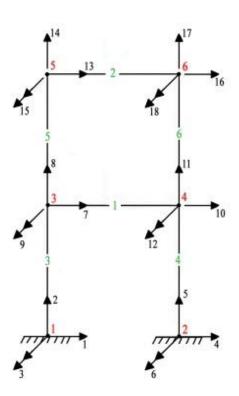


Fig. 1 Members and Nodes Numbers

1. Load Vector - Find out the load vector (member fixed end forces) for each member , for this example member length 4 m.

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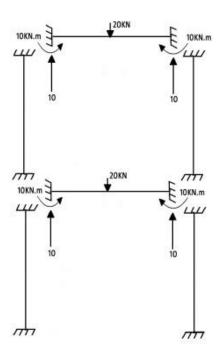


Fig.2 Fixed end action due to external load

**2. Stiffness matrix -** Determine the load vector due to joint displacements (i.e. member stiffness equations) for each member.

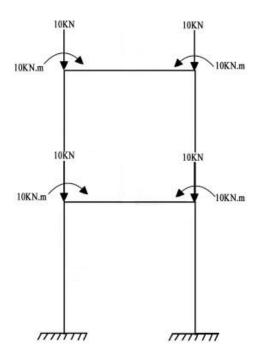


Fig.3 Equivalent joint loads

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- **3. Equilibrium equation -** Under the influences of forces in previous two steps and joint loads, write the equations of equilibrium of all joints.
- **4. Solving equilibrium equation -** Solve equilibrium equation to find the joint displacements.

$$[P] = [K] [\Delta]$$

**5. Internal forces** - By using the resulting joint displacements, determine the total member end forces.

#### IV.DESCRIPTION OF THE PROBLEM

**Example 1**: Analyse the Multistorey rigit frame of G+3, one bay with point load 50 KN on all horizontal elements, shown in Fig.4 by stiffness matrix method. Assume E=200 Gpa;  $I=1.33\times10^{-5}$  m<sup>4</sup> and A=0.01 m<sup>2</sup>. The flexural rigidity EI and axial rigidity EA are the same for all beams.

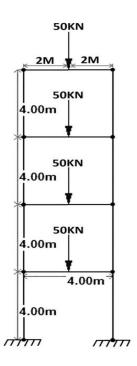


Fig. 4 Multistorey Frame, G+4 one bay with point loads

The plane frame is divided in to twelve beam elements as shown in Fig. 4 the numbering of joints and members are also shown in Fig. 5 and Fig. 6 the possible degrees of freedom at nodes are also shown in the figure. The origin of the global co- ordinate system is taken at node no. 1.

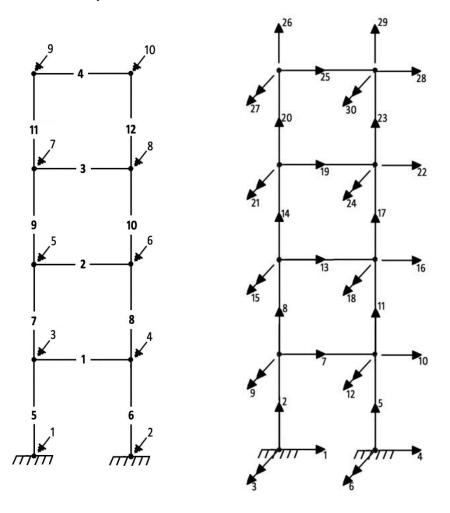


Fig. 5 Node and Member no.

Fig. 6 Degree of freedom on each node.

 $\label{eq:Table-1} Table-1$  Nodes Deflection Values of G+3

Node no.	Degree of	Degree of	Manual Calculation	STAAD Pro.	Deference	Units
rvode no.	freedom	freedom no.	Withing Carculation	Calculation	Belefeliee	
	X	7	-8.22×10 <sup>-4</sup>	-0.003	0.002	mm
3	Y	8	-0.200	-0.200	0	mm
	Mz	9	-0.003	-0.003	0	Rad
	X	10	0.0045	0.003	0.001	mm

4	Y	11	-0.2221	-0.200	0.022	mm
4						
	Mz	12	0.003	0.003	0	Rad
	X	13	0.0215	0.001	0.02	mm
5	Y	14	-0.350	-0.350	0	mm
	Mz	15	-0.0027	-0.003	0.001	Rad
	X	16	0.0193	-0.001	0.018	mm
6	Y	17	-0.394	-0.350	0.044	mm
	Mz	18	0.0026	0.003	0.0004	Rad
	X	19	0.053	-0.003	0.05	mm
7	Y	20	-0.450	-0.450	0	mm
	Mz	21	-0.002	-0.002	0	Rad
	X	22	0.058	0.003	0.055	mm
8	Y	23	-0.516	-0.450	0.066	mm
	Mz	24	0.002	0.002	0	Rad
	X	25	0.080	0.008	0.072	mm
9	Y	26	-0.500	-0.500	0	mm
	Mz	27	-0.005	-0.005	0	Rad
	X	28	0.077	-0.008	0.069	mm
10	Y	29	-0.588	-0.500	0.088	mm
	Mz	30	0.005	0.006	0.001	Rad

 $\label{eq:Table-2} Table-2$  Reaction Values of Foundation Nodes of G+3

Node no.	Degree of freedom	Reaction no.	Manual Calculation	STAAD Pro. Calculation	Deference	Units
	X	$R_1$	3.200	3.211	0.011	KN
1	Y	$R_2$	100.050	100.000	0.050	KN
1	Mz	$R_3$	-4.256	-4.276	0.020	KNm
	X	$R_4$	-3.202	-3.211	0.090	KN
2	Y	$R_5$	100.000	100.000	0	KN
	Mz	$R_6$	4.260	4.276	0.016	KNm

 $\label{eq:Table-3} Table-3$  Element End Banding Moments of G+3

Element	Node	Degree of	Manual	STAAD Pro.	Deference
no.	no.	freedom	(KNm)	(KNm)	Deference
1	N <sub>3</sub>	Mz	20.74	20.70	0.04
_	N <sub>4</sub>	Mz	-20.74	-20.70	0.04
2	$N_5$	Mz	21.42	21.40	0.02
	$N_6$	Mz	-21.40	-21.40	0
3	$N_7$	Mz	22.21	22.18	0.03
	$N_8$	Mz	-22.17	-22.18	0.02
4	$N_9$	Mz	17.67	17.58	0.09
4	$N_{10}$	Mz	-17.52	-17.58	0.06
5	$N_1$	Mz	-4.256	-4.276	
3	N <sub>3</sub>	Mz	-8.512	-8.569	0.02
6	N <sub>2</sub>	Mz	4.256	4.276	0
0	N <sub>4</sub>	Mz	8.512	8.569	0.02
7	$N_3$	Mz	-12.103	-12.138	0
′	$N_5$	Mz	-11.438	-11.436	0.02
8	N <sub>4</sub>	Mz	12.103	12.138	0
0	$N_6$	Mz	11.438	11.436	0.02
9	$N_5$	Mz	-9.87	-9.973	0
7	$N_7$	Mz	-9.07	-9.197	0.02
10	N <sub>6</sub>	Mz	10.075	9.973	0
10	N <sub>8</sub>	Mz	9.277	9.197	0.02
11	N <sub>7</sub>	Mz	-13.034	-12.988	0
11	N <sub>9</sub>	Mz	-17.68	-17.586	0.02
12	N <sub>8</sub>	Mz	13.034	12.988	0
12	N <sub>10</sub>	Mz	17.68	17.586	0.02

 $\label{eq:Table-4} Table-4$  Element End Shear Forces of G+3

Element	Node	Degree of	Manual	STAAD Pro.	Deference
no.	no.	freedom	(KN)	(KN)	Deference
1	$N_3$	Y	25	25	0
	N <sub>4</sub>	Y	25	25	0
2	N <sub>5</sub>	Y	25.005	25	0.005
	N <sub>6</sub>	Y	24.995	25	0.005
3	N <sub>7</sub>	Y	25.01	25	0.03
	$N_8$	Y	24.98	25	0.02
4	N <sub>9</sub>	Y	25.03	25	0.09
4	$N_{10}$	Y	24.96	25	0.06
5	$N_1$	Y	-3.200	-3.211	0.011
3	$N_3$	Y	3.2	3.211	0.011
6	$N_2$	Y	3.200	3.211	0.011
0	$N_4$	Y	-3.2	-3.211	0.011
7	$N_3$	Y	-5.90	-5.893	0.007
,	$N_5$	Y	5.9	5.893	0.007
8	$N_4$	Y	5.90	5.893	0.007
0	$N_6$	Y	-5.9	-5.893	0.007
9	$N_5$	Y	-4.75	-4.792	0.043
	$N_7$	Y	4.75	4.792	0.042
10	$N_6$	Y	4.75	4.792	0.042
10	$N_8$	Y	-4.85	-4.792	0.058
11	$N_7$	Y	-7.70	-7.643	0.057
	N <sub>9</sub>	Y	7.70	7.643	0.057
12	$N_8$	Y	7.70	7.643	0.057
12	N <sub>10</sub>	Y	-7.70	-7.643	0237

**Example 2:** Analyse the Multistorey rigit frame of G+5, one bay with point load 40 KN and UDL 20KN/m on all horizontal elements, shown in Fig.7 by stiffness matrix method. Assume E=200 Gpa;  $I = 1.33 \times 10^{-5}$  m<sup>4</sup> and A=0.01 m<sup>2</sup>. The flexural rigidity EI and axial rigidity EA are the same for all beams.

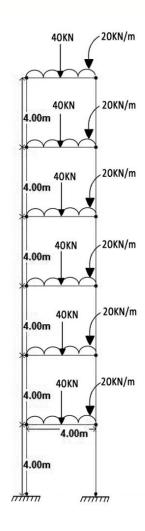


Fig.7 Multistorey Frame G+5 One bay with point load & U.D.L.

 $\label{eq:Table-5} Table-5$  Nodes Deflection Values of G+5

Node	Degree of	Deflectio	Manual	STAAD Pro.	Deference	Units
no.	freedom	n no.		Calculation		
	X	$\mathbf{u}_7$	-0.0048	-0.005	0.0002	mm
3	Y	$u_8$	-0.720	-0.702	0.018	mm
	Mz	<b>u</b> 9	-0.006	-0.006	0	Rad
	X	$u_{10}$	0.0048	0.005	0.0002	mm
4	Y	$u_{11}$	-0.720	-0.702	0.018	mm
	Mz	$u_{12}$	0.006	0.006	0	Rad
	X	u <sub>13</sub>	0.001	0.001	0	mm

5	Y	u <sub>14</sub>	-1.320	-1.288	0.032	mm
	Mz	u <sub>15</sub>	-0.0048	-0.005	0.0002	Rad
	X	u <sub>16</sub>	-0.001	-0.001	0	mm
6	Y	u <sub>17</sub>	-1.320	-1.288	0.032	mm
	Mz	u <sub>18</sub>	0.0048	0.005	0.0002	Rad
	X	u <sub>19</sub>	-0.0004	-0.000	0.0004	mm
7	Y	u <sub>20</sub>	-1.800	-1.756	0.044	mm
	Mz	u <sub>21</sub>	-0.005	-0.005	0	Rad
	X	u <sub>22</sub>	0.0004	0.000	0.0004	mm
8	Y	u <sub>23</sub>	-1.800	-1.756	0.044	mm
	Mz	u <sub>24</sub>	0.005	0.005	0	Rad
	X	u <sub>25</sub>	0.001	0.001	0	mm
9	Y	u <sub>26</sub>	-2.160	-2.107	0.053	mm
	Mz	u <sub>27</sub>	-0.005	-0.005	0	Rad
	X	u <sub>28</sub>	-0.001	-0.001	0	mm
10	Y	u <sub>29</sub>	-2.160	-2.107	0.053	mm
	Mz	u <sub>30</sub>	0.005	0.005	0	Rad
	X	u <sub>31</sub>	-0.005	-0.005	0	mm
11	Y	u <sub>32</sub>	-2.400	-2.341	0.059	mm
11	Mz	u <sub>33</sub>	-0.0039	-0.004	0.0001	Rad
	X	u <sub>34</sub>	0.005	0.005	0	mm
12	Y	u <sub>35</sub>	-2.400	-2.341	0.059	mm
12	Mz	u <sub>36</sub>	0.0039	0.004	0.0001	Rad
	X	u <sub>37</sub>	0.014	0.014	0	mm
13	Y	u <sub>38</sub>	-2.520	-2.459	0.061	mm
13	Mz	u <sub>39</sub>	-0.01	-0.01	0	Rad
	X	u <sub>40</sub>	-0.014	-0.014	0	mm
14	Y	$u_{41}$	-2.520	-2.459	0.061	mm
17	Mz	u <sub>42</sub>	0.010	0.010	0	Rad

 $\label{eq:Table-6} Table-6$  Reaction Values of Foundation Nodes of G+5

Node	Degree of	Reaction	Manual Calculation	STAAD Pro.	Deference	Units	
no.	freedom	no.		Calculation	Deference	Omis	
	X	$R_1$	6.059	6.038	0.021	KN	
1	Y	$R_2$	360	360	0	KN	
	Mz	$R_3$	-8.060	-8.040	0.020	KNm	
	X	R <sub>4</sub>	-6.0591	-6.038	0.021	KN	
2	Y	$R_5$	360	360	0	KN	
	Mz	$R_6$	8.0602	8.040	0.02	KNm	

 $\label{eq:Table-7} Table-7$  Element End Banding Moments of G+5

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Element	Node	Degree of	Manual Calculation	STAAD Pro.  Calculation	Deference
no.	no.	freedom	(KNm)	(KNm)	201010110
1	$N_3$	Mz	38.6113	38.594	0.017
	N <sub>4</sub>	Mz	-38.6113	-38.594	0.017
2	$N_5$	Mz	40.276	40.258	0.018
	$N_6$	Mz	-40.276	-40.258	0.018
3	$N_7$	Mz	40.0044	39.987	0.017
	$N_8$	Mz	-40.0044	-39.987	0.017
4	N <sub>9</sub>	Mz	39.7019	39.686	0.015
4	N <sub>10</sub>	Mz	-39.7019	-39.686	0.015
5	N <sub>11</sub>	Mz	-41.491	-41.470	0.021
3	N <sub>12</sub>	Mz	-41.491	-41.470	0.021
6	N <sub>13</sub>	Mz	32.8298	32.808	0.021
0	N <sub>14</sub>	Mz	-32.8298	-32.808	0.021
7	$N_1$	Mz	-8.0602	-8.040	0.020
,	N <sub>3</sub>	Mz	-16.1156	-16.113	0.002
8	N <sub>2</sub>	Mz	8.0602	8.040	0.020
o	$N_4$	Mz	16.1156	16.113	0.002
9	$N_3$	Mz	-22.4956	-22.482	0.013
7	N <sub>5</sub>	Mz	-20.8309	-20.818	0.012
10	$N_4$	Mz	22.4956	22.482	0.013

	$N_6$	Mz	20.8309	20.818	0.012
11	N <sub>5</sub>	Mz	-19.4451	-19.439	0.006
11	N <sub>7</sub>	Mz	-19.7167	-19.710	0.006
12	N <sub>6</sub>	Mz	19.4450	19.439	0.006
12	N <sub>8</sub>	Mz	19.7167	19.710	0.006
13	N <sub>7</sub>	Mz	-20.2878	-20.277	0.010
13	N <sub>9</sub>	Mz	-20.5903	-20.577	0.013
14	N <sub>8</sub>	Mz	20.2878	20.277	0.010
14	N <sub>10</sub>	Mz	20.5903	20.577	0.013
15	N <sub>9</sub>	Mz	-19.1115	-19.109	0.002
13	N <sub>11</sub>	Mz	-17.3224	-17.324	0.001
16	N <sub>10</sub>	Mz	19.1115	19.109	0.002
10	N <sub>12</sub>	Mz	17.3224	17.324	0.002
17	N <sub>11</sub>	Mz	-24.1687	-24.146	0.022
1 /	N <sub>13</sub>	Mz	-32.8298	-32.808	0.021
18	N <sub>12</sub>	Mz	24.1687	24.146	0.022
10	N <sub>14</sub>	Mz	32.8298	32.808	0.021

 $\label{eq:Table-8} Table-8$  Element End Shear Forces of G+5

Element	Node	Degree of	Manual	STAAD Pro.	
no.		freedom	Calculation	Calculation	Deference
no.	no.	needom	(KN)	(KN)	
1	$N_3$	Y	60	60	0
	$N_4$	Y	60	60	0
2	$N_5$	Y	60	60	0
	$N_6$	Y	60	60	0
3	$N_7$	Y	60	60	0
	$N_8$	Y	60	60	0
4	$N_9$	Y	60	60	0
	$N_{10}$	Y	60	60	0
5	N <sub>11</sub>	Y	60	60	0
	N <sub>12</sub>	Y	60	60	0
6	N <sub>13</sub>	Y	60	60	0
	N <sub>14</sub>	Y	60	60	0

7	$N_1$	Y	-6.0591	-6.038	0.021
,	$N_3$	Y	6.0591	6.038	0.021
8	$N_2$	Y	6.0591	6.038	0.021
8	$N_4$	Y	-6.0591	-6.038	0.021
9	$N_3$	Y	-10.8588	-10.825	0.033
9	N <sub>5</sub>	Y	10.8588	10.825	0.033
10	N <sub>4</sub>	Y	10.8588	10.825	0.033
10	N <sub>6</sub>	Y	-10.8588	-10.825	0.033
11	$N_5$	Y	-9.8150	-9.787	0.028
11	N <sub>7</sub>	Y	9.8150	9.787	0.028
12	N <sub>6</sub>	Y	9.8150	9.787	0.028
12	N <sub>8</sub>	Y	-9.8150	-9.787	0.028
13	$N_7$	Y	-10.2451	-10.213	0.032
13	N <sub>9</sub>	Y	10.2451	10.213	0.032
14	N <sub>8</sub>	Y	10.2451	10.213	0.032
14	N <sub>10</sub>	Y	-10.2451	-10.213	0.032
15	N <sub>9</sub>	Y	-9.1313	-9.108	0.023
13	N <sub>11</sub>	Y	9.1313	9.108	0.023
16	N <sub>10</sub>	Y	9.1313	9.108	0.023
10	N <sub>12</sub>	Y	-9.1313	-9.108	0.023
17	N <sub>11</sub>	Y	-14.2854	-14.238	0.047
1/	N <sub>13</sub>	Y	14.2854	14.238	0.047
18	N <sub>12</sub>	Y	14.2854	14.238	0.047
10	N <sub>14</sub>	Y	-14.2854	-14.238	0.047

#### IV.RESULTS AND DISCUSSIONS

The first part of the work is completing of manual analysis using Stiffness Matrix Method and Second part is analysis in STAAD Pro software mostly used for analysis and design, and third part is comparison the results of both analysis. The same frame which is chosen for the research in model in STAAD Pro with applied the loading which is calculated by manual means. The compare final results of vertical and horizontal elements of stiffness matrix method with STAAD Pro results then only maximum deference is 0.09 in results.

#### **V.CONCLUSION**

The manual analysis by stiffness matrix method and STAAD Pro analysis, the following conclusion can be listed follows:

- 1. Stiffness matrix method is easy and full with lots off number as a degree of freedom in matrix of 30×30 for G+3 and 42×42 for G+5 subjected to vertical loadings of point load and U.D.L.
- 2. The comparison of manual and software's results, the results are quite same.
- 3. STAAD Pro is a great software for analysis and design, the only high skilled parson can use but we are creating Excel Sheets on base of this manual calculation, it will helps to every medal class constructors for saving time and money as we know STAAD Pro software and its License is costly in market.
- 4. The Excel sheets can be recommended for analysis of standard and mostly used for construction portal frames Etc.

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